# Lecture 16: Gravitational potential energy and space travel

- Universal gravitational potential energy
- Space travel problems
- Escape speed
- Orbital energy
- Multiple objects

### Gravitational potential energy

$$F_{grav} = \frac{GmM}{r^2}$$
, attractive Conservative force

$$U_B - U_A = -W_{A \to B} = -\int_{\vec{r}_A}^{\vec{r}_B} \vec{F} \cdot d\vec{r}$$

$$\begin{split} U_B - U_A &= -W_{A \to B} = -\int_{\vec{r}_A}^{\vec{r}_B} \vec{F} \cdot d\vec{r} \\ \int_{\vec{r}_A}^{\vec{r}_B} \vec{F} \cdot d\vec{r} &= \int_{r_A}^{r_B} F_r dr = \int_{r_A}^{r_B} -\frac{GmM}{r^2} dr \\ &= -\left[-\frac{GmM}{r}\right]_{r_B}^{r_B} = \frac{GmM}{r_B} - \frac{GmM}{r_A} \end{split}$$

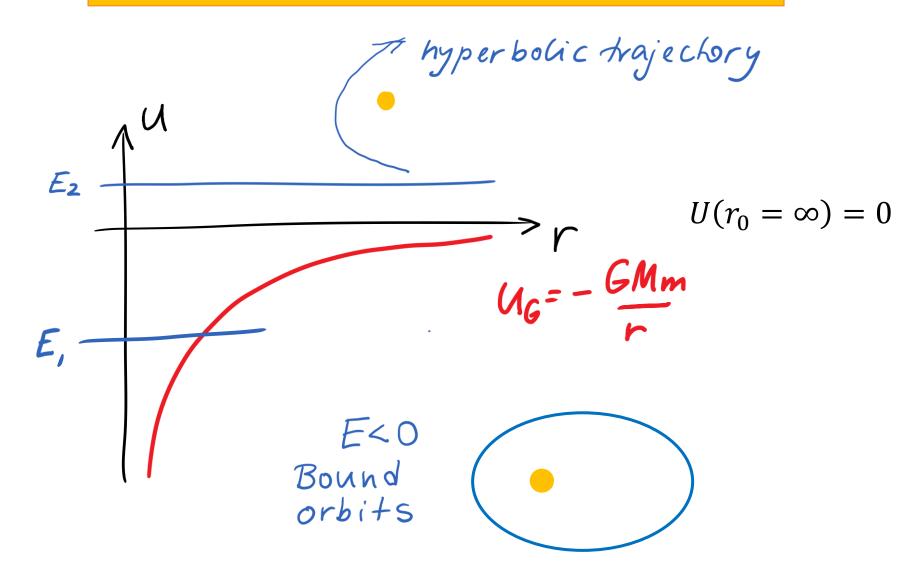
$$U_B - U_A = -GmM\left(\frac{1}{r_B} - \frac{1}{r_A}\right)$$

Choose reference point:  $r_0 = \infty$ . Assign  $U(r_0 = \infty) = 0$ 

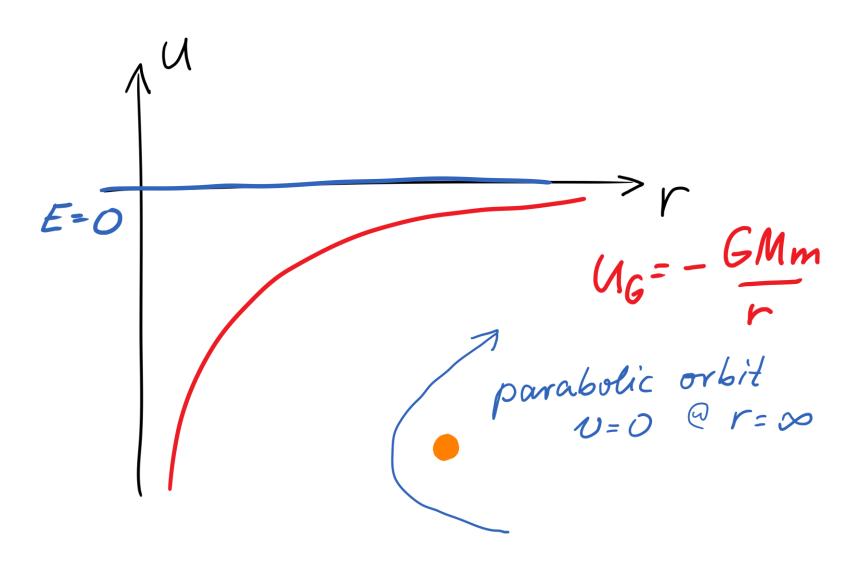
$$U_{Grav} = -\frac{GmM}{r}$$

negative!

#### Potential energy diagram



# Potential energy diagram



#### **Escape condition**

$$E = \frac{1}{2}mv^2 - \frac{GmM}{r}$$

$$V_i = 0$$

$$V_i = R$$

Critical escape condition: barely making it to r = infinity, having slowed to speed of zero

#### Escape speed

Minimum speed an object must have at distance *R* from central mass *M* if it is to go infinitely far away

$$E_i = E_f$$

$$\frac{1}{2}mv_{esc}^2 - \frac{GmM}{R} = \frac{1}{2}mv_{\infty}^2 - \frac{GmM}{r = \infty}$$

$$\frac{1}{2}mv_{esc}^2 - \frac{GmM}{R} = 0 - 0$$

$$v_{esc} = \sqrt{\frac{2GM}{R}}$$

,

# Example: Escape speed from Earth

$$M_{Earth} = 5.97 \times 10^{24} kg$$
  
 $R_{Earth} = 6.38 \times 10^6 m$ 

$$G = 6.67 \times 10^{-11} N m^2/kg^2$$

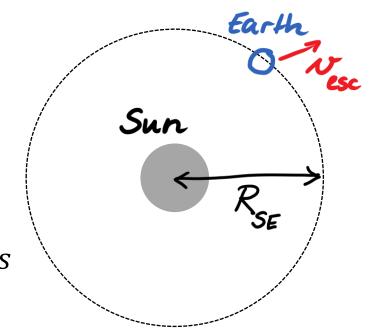
$$v_{esc} = \sqrt{\frac{2GM}{R}} = 11,200 \text{ m/s}$$

#### Example: Escape speed from orbit

Speed necessary to escape gravitational field of the Sun when object is launched from Earth:

$$M_{Sun}$$
=1.99 × 10 <sup>30</sup> kg  
 $R_{SE}$ = 1.50 × 10 <sup>11</sup> m  
 $G$ = 6.67 × 10 <sup>-11</sup> N m<sup>2</sup>/kg<sup>2</sup>

$$v_{esc} = \sqrt{\frac{2GM_{Sun}}{R_{SE}}} = 42,000 \, m/s$$



## **Orbital Energy**

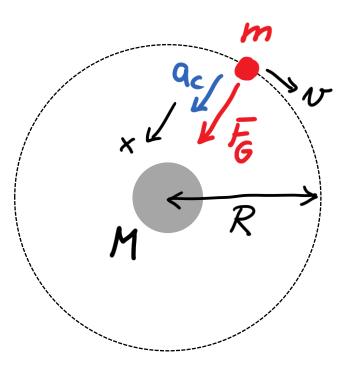
$$E = K + U = \frac{1}{2}mv^2 - \frac{GmM}{R}$$

Speed of satellite in circular orbit:  $v^2 = \frac{GM}{R}$ 

$$E = \frac{1}{2}m\left(\frac{GM}{R}\right) - \frac{GmM}{R}$$

$$E = -\frac{GmM}{2R}$$



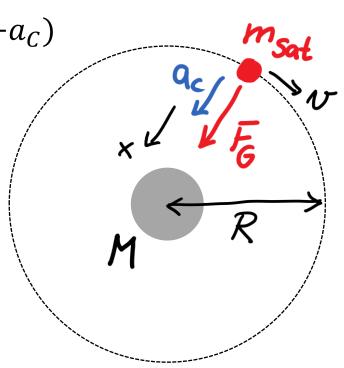


#### **Satellite Motion**

$$\sum F_{x} = F_{grav,x} = m_{sat}a_{x} = m_{sat}(+a_{c})$$

$$\frac{Gm_{sat}M}{R^2} = m_{sat}\frac{v^2}{R}$$

$$\frac{GM}{R} = v^2$$



# Multiple objects

$$F_{2}, \frac{M_{2}}{F_{2}}$$

$$F_{3}, \frac{M_{2}}{F_{3}}$$

$$F_{4}, \frac{M_{2}}{F_{4}}$$

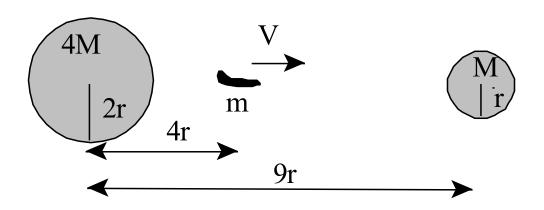
$$F_{5}, \frac{M_{2}}{F_{5}}$$

$$F_{5}, \frac{M_{2}}{F_{$$

#### Example with multiple objects

A planet has mass 4M and a radius 2r. Its moon has mass M and radius r. The centers of the planet and moon are a distance 9r apart.

A shuttle of mass m is a distance 4r away from the center of the planet and moving with speed V. What is the total mechanical energy of the shuttle?



If the shuttle was initially at rest at X, how much work did the engines do?

